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Finite-element formulation of a nonlocal hereditary fractional-order Timoshenko beam

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*Original*

Finite-element formulation of a nonlocal hereditary fractional-order Timoshenko beam / Alotta, G., Failla, G., Zingales, M. - In: JOURNAL OF ENGINEERING MECHANICS. - ISSN 0733-9399. - 143:5(2017), p. D4015001.D4015001. [10.1061/(ASCE)EM.1943-7889.0001035]

*Availability:*

This version is available at: <https://hdl.handle.net/20.500.12318/1960> since: 2021-01-26T16:55:09Z

*Published*

DOI: [http://doi.org/10.1061/\(ASCE\)EM.1943-7889.0001035](http://doi.org/10.1061/(ASCE)EM.1943-7889.0001035)

The final published version is available online at: [https://ascelibrary.org/doi/abs/10.1061/\(ASCE\)EM.1943-](https://ascelibrary.org/doi/abs/10.1061/(ASCE)EM.1943-)

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# FINITE ELEMENT FORMULATION OF A NON-LOCAL HEREDITARY FRACTIONAL-ORDER TIMOSHENKO BEAM

Gioacchino Alotta\*, Giuseppe Failla\*\*, Massimiliano Zingales\*\*\*

## ABSTRACT

A mechanically-based non-local Timoshenko beam model, recently proposed by the authors, hinges on the assumption that non-local effects can be modeled as elastic long-range volume forces and moments mutually exerted by non-adjacent beam segments, which contribute to the equilibrium of any beam segment along with the classical local stress resultants. Long-range volume forces/moments linearly depend on the product of the volumes of the interacting beam segments, and on pure deformation modes of the beam, through attenuation functions governing the space decay of non-local effects.

This paper investigates the response of this non-local beam model when viscoelastic long-range interactions are included, modeled by Caputo's fractional derivatives. The finite element method is used to discretize the pertinent fractional-order equations of motion. Closed-form solutions are

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17 obtained for creep tests by typical tools of fractional calculus. Numerical results are presented for  
18 various non-local parameters.

19

## 20 **KEYWORDS**

21 Non-local Viscoelasticity; Non-local Damping; Fractional Calculus; Long-range Interactions;  
22 Timoshenko Beam.

23

## 24 **INTRODUCTION**

25 In the last few decades, much effort has been devoted to develop non-local beam theories.  
26 Certainly, one of the reasons is the need for adequate and computationally-efficient modeling of  
27 microstructural effects in beam-like micro- and nano-devices (Lakes 1991; Aifantis 1994; Qian et  
28 al. 2002; Arash and Wang 2012). Indeed, these effects, which have been revealed by experimental  
29 tests on materials such as graphite (Tang 1983), copper (Poole et al. 1996), epoxy (Lam et al. 2003)  
30 and polypropylene (McFarland and Colton 2005), cannot be described by the intrinsically free-scale  
31 classical continuum approach while, on the other hand, could be captured only at the expense of  
32 computationally intensive and, in some cases, almost prohibitive atomistic/molecular simulations  
33 (Wang and Hu 2005). A further important application of non-local beam theories is at a  
34 macroscopic scale, whenever an intrinsic dependence exists between the response at a given point  
35 and the response at surrounding points of a beam. Such a dependence may arise as a result of  
36 external patches, long adhesive joints in composites, surface treatments using fluids, or fibers in  
37 fiber-reinforced composites. In these cases, instead of modeling all components of the system, as  
38 beam and external patch, or composite matrix and embedded fibers, a simpler yet accurate solution  
39 can be obtained from 1D equilibrium equations of the beam, where coupling between responses at  
40 non-adjacent points is accounted for by appropriate non-local terms. Non-local beam theories are  
41 also suitable for modeling effects produced, at a given point, by the complex deformations of non-

42 adjacent beam cross sections, as these effects cannot be captured by classical beam models where  
43 cross section remain planes (Lei et al. 2006; Challamel 2011, 2013).

44 In general, non-local beam theories rely on introducing non-local terms in a classical continuum,  
45 which is formulated within the framework of classical Euler-Bernoulli (EB), Timoshenko (TM) or  
46 higher-order beam theories. Following this approach, early non-local beam models have been built  
47 using the non-local Eringen's integral law for normal and shear stress (Eringen 1972, 1983) in EB  
48 and TM beam models (e.g., see the study by Lu et al. (2007) and references therein), and higher-  
49 order beam models (Reddy 2007; Aydogdu 2009). Also, several non-local theories alternative to  
50 Eringen's integral theory have been used to build non-local beam models. Among the many, there  
51 exist non-local EB beam models based on modified couple stress theories (Park and Gao 2006;  
52 Kong et al. 2008), general strain gradient elasticity theory (Lam et al. 2003), gradient elasticity  
53 theory and integral elasticity theory with a constitutive relation combining local and non-local  
54 curvatures (Challamel and Wang 2008), micropolar elasticity constitutive law (McFarland and  
55 Colton 2005), and a hybrid approach involving a strain energy functional with local and non-local  
56 curvatures (Zhang et al. 2010). Non-local TM beam models have been built by Wang et al. (2010)  
57 in conjunction with the strain gradient elasticity theory presented by Lam et al. (2003), by Ma et al.  
58 (2008) based on a modified couple stress theory. Also, non-local EB and TM beam models have  
59 been developed based on a stress gradient elasticity theory (Pradhan 2012; Yang and Lim 2012).  
60 Very recently, non-local EB and TM beam models have been proposed by fractional generalizations  
61 of gradient elasticity theories, based on a new fractional variational principle for Lagrangians with  
62 Riesz fractional derivatives (Tarasov and Aifantis 2015).

63 In the non-local beam models briefly recalled above, non-locality affects the stiffness terms.  
64 However, an interesting and challenging task is a non-local modeling of damping effects. In fact,  
65 non-local damping models could be of interest to capture damping effects at a microstructural level  
66 that, as recent studies show, may play an important role in image acquisition via high-speed  
67 atomic force microscopes as the scan rates increase (Payton et al., 2012), or may significantly affect

68 the frequency measurements of vibrating nano-sensors (Murmu and Adhikari, 2012) detecting the  
69 mass of small particles based on shifts in measured frequencies (Calleja et al., 2012). Also, damping  
70 effects in nanostructures have been detected as a result of external magnetic forces (Lee and Lin,  
71 2010), humidity or thermal effects (Chen et al., 2011). At a macroscopic level, on the other hand,  
72 non-local damping may be produced when responses at non-adjacent points are coupled by external  
73 patches, adhesive joints or surface treatments, or by embedded fibers in composites (Lei et al. 2006;  
74 Friswell et al. 2007).

75 In order to model damping effects in nano-beams, EB beam models with non-local viscoelastic  
76 behavior have been proposed by Lei et al. (2013), including multi-parameter time-dependent  
77 viscoelastic terms in the standard Eringen's law for normal stress. Non-local Kelvin-Voigt and  
78 three-parameter viscoelastic models have been discussed in detail. Applications have been  
79 presented on single walled carbon nanotubes, using a transfer function approach for free vibration  
80 analysis. The torsional behavior of functionally-graded nano-beams, including non-local  
81 viscoelasticity by suitable modifications of Eringen's law for shear stress, has been studied by  
82 Barretta et al. (2015) and, in particular, a closed-form response has been obtained for a viscoelastic  
83 model including a Maxwell model connected in series with a Voigt model.

84 As for non-local damping effects at a macroscopic level, an early EB beam model with non-local  
85 damping has been proposed by Russell (1992). Equations of motion include a viscous long-range  
86 moment per unit length, given by an integral depending on the relative rate of rotation between the  
87 beam segment and non-adjacent ones, through an appropriate attenuation function. An additional  
88 moment is also involved in the natural boundary conditions (B.C.). The model was conceived for  
89 EB composite beams with longitudinal fibers, such as fiberglass, boron and graphite composites, to  
90 account for the dissipation that may occur at the fiber-matrix interface due to imperfect bonding.  
91 Russell (1992) modeled the effects of such a dissipation as a damping moment, which depends on  
92 the relative rate of rotation between non-adjacent beam segments, in recognition of the fact that,  
93 while a differential rotation takes place along the beam axis, opposite motions of the fibers relative

94 to the matrix occur above and below the elastic axis, and the resulting dissipation forces transmitted  
95 from the fibers to the matrix produce indeed a damping moment, within any beam segment which  
96 the fibers pass through. Obviously, the attenuation function reflects that coupling due to fibers  
97 progressively decays with distance. Experimental evidence for this non-local damping model has  
98 been found by Russell (1992) in the free vibrations of a boron-epoxy composite beam and, later, by  
99 Banks and coworkers (Banks and Inman 1991; Banks et al. 1994). A similar damping mechanism  
100 has been proposed by Russell (1992) also for the longitudinal vibrations of a fiber-reinforced  
101 composite bar.

102 More recently, Friswell and coworkers (Lei et al. 2006; Friswell et al. 2007) have proposed a  
103 non-local EB beam model where non-local damping terms are built as a weighted average of a  
104 velocity field over the beam domain, with appropriate attenuation functions taken as weighting  
105 functions. External and internal non-local damping models have been considered, depending on the  
106 transverse displacement and its fourth-order derivative, respectively. While the external damping  
107 model is seen as the result of external damping patches, long adhesive joints in composites or  
108 surface damping treatments using fluids, the internal non-local damping model has been thought as  
109 a homogenized model of an intrinsic dependence between the response at a given point and the  
110 response at the surrounding points of the medium (Flügge 1975). Such a dependence may be  
111 associated with effects produced at a given point by the complex deformations of non-adjacent  
112 beam cross sections, not adequately described by the plane-section assumption of classical beam  
113 models (Lei et al. 2006) or, alternatively, may compensate for uncertainties in spatial location of the  
114 damping sources, and dependence of damping mechanism on the material microstructure (Friswell  
115 et al. 2007). In their work, Friswell and coworkers have considered either viscous or viscoelastic  
116 non-local damping, the latter with time-dependent exponential forms (Friswell et al. 2007).

117 In the last few years, the authors have proposed non-local EB and TM beam models (Di Paola et  
118 al. 2013, Di Paola et al. 2014, Alotta et al. 2014, Failla et al. 2015), within a mechanically-based  
119 approach to non-locality, which treats non-local effects as long-range interactions resulting from

120 relative motion of non-adjacent volume elements (Di Paola et al. 2009, 2010a, 2010b; Failla et al.  
121 2010, 2013). In these non-local beam models, in particular, long-range interactions are volume  
122 forces/moments resulting from a differential motion of non-adjacent beam segments, measured by  
123 the pure deformation modes of the beam (Fuchs 1991, 1997), i.e. a “pure axial” symmetric mode, a  
124 “pure bending” symmetric mode and a “pure shear” asymmetric mode. The analytical form of the  
125 long-range volume forces/moments is built as linearly depending on the product of the volumes of  
126 the interacting beam segments, and the pure deformation modes, through pertinent attenuation  
127 functions governing the space decay of the non-local effects. In previous studies, the authors have  
128 considered elastic and viscous long-range interactions, either separately or simultaneously (Di Paola  
129 et al. 2013, Di Paola et al. 2014, Alotta et al. 2014, Failla et al. 2015).

130 In this paper, the purpose is to re-formulate the non-local TM beam model previously proposed  
131 by the authors, in order to include fractional-order viscoelastic long-range interactions. Fractional  
132 derivatives are indeed well-recognized mathematical tools for modeling long-memory effects  
133 (Tarasov and Zaslavsky 2007, 2008), and have already proved particularly suitable for modeling  
134 viscoelastic behavior (Rabotnov 1980, Bagley and Torvik 1983a, 1983b, 1985, 1986; Mainardi  
135 2010; Meral et al. 2010; Di Paola et al. 2011; Di Paola et al. 2012; Di Paola et al. 2014; Failla and  
136 Pirrotta 2012, Sapora et al. 2014, Scimemi and Ponte 2014; Di Lorenzo et al. 2014). Here, in  
137 particular, the Caputo’s fractional derivative will be used (Podlubny 1999) to model the fractional-  
138 order long-range interactions. On deriving the equations of motion, a corresponding discrete form  
139 will be obtained by the finite element (FE) method. Then, it will be shown that closed-form  
140 solutions may readily be derived for creep tests by simple rules of fractional calculus.

141 The paper is organized as follows. After a brief description of fractional operators in Section 2,  
142 the non-local TM beam model is introduced in Section 3, and related equations of motion in Section  
143 4. FE discretization is described in Section 5, while closed-form solutions for creep tests are  
144 presented in Section 6. Numerical applications are discussed in Section 7.

146 **FRACTIONAL-ORDER HEREDITARINESS**

147 Preliminary investigations on fractional derivatives as applied to viscoelasticity modeling trace back  
148 to the work of Gemant (1938) and Bosworth (1946), who were the first to propose a fractional  
149 derivative model for viscoelasticity, and the studies by Scott-Blair and Gaffyn (1949) and Caputo  
150 (1974), who fitted fractional derivatives to experimental data. Later, Bagley and Torvik (1983a,  
151 1983b, 1985, 1986) framed a fractional derivative viscoelastic model in the context of molecular  
152 theory, showing that, in order to capture the frequency-dependence of damping properties in  
153 viscoelastic materials, fractional derivatives are more appropriate than classical linear models such  
154 as the Kelvin–Voigt model. In the last three decades a considerable number of studies (Rogers  
155 1983; Koeller 1984; Pritz 1996; Galucio et al. 2004; Adolfsson et al. 2005) have substantiated the  
156 capability of fractional derivatives to describe complex viscoelastic material behavior, in form of  
157 equations involving a small number of parameters (Di Paola et al. 2011).

158 Several viscoelastic models are based on the following Caputo’s definition of fractional  
159 derivative (Podlubny 1999):

160

$${}_c D_{0^+}^\alpha f(t) = \frac{1}{\Gamma(1-\alpha)} \int_0^t \frac{1}{(t-\tau)^\alpha} \frac{df(\tau)}{d\tau} d\tau \quad 0 < \alpha < 1 \quad (1)$$

161

162 It can readily be seen that, for systems at rest at  $t = 0$  the Caputo’s fractional derivative coincides  
163 with the Riemann-Liouville fractional derivative, defined as (Podlubny 1999):

164

$${}_{RL} D_{0^+}^\alpha f(t) = \frac{1}{\Gamma(1-\alpha)} \frac{d}{dt} \int_0^t \frac{1}{(t-\tau)^\alpha} f(\tau) d\tau \quad 0 < \alpha < 1 \quad (2)$$

165

166 Time-domain discretization of the Caputo’s fractional derivative (1) can be made by the well-  
167 known Grunwald-Letnikov algorithm (Spanos and Evangelatos 2010):

$$(D_C^\alpha x)(t) = \lim_{\Delta t \rightarrow 0} \Delta t^{-\alpha} \sum_{k=0}^i GL_k x(t_i - k\Delta t) \quad (3)$$

168

169 where  $GL_k$  are coefficients to be computed in the recursive form

170

$$GL_k = \frac{k - \alpha - 1}{k} GL_{k-1}, \quad GL_0 = 1.0 \quad (4)$$

171

## 172 BEAM MODEL

173 Figure 1 shows a beam of arbitrary cross section, referred to a Cartesian (orthogonal) coordinate  
 174 system  $Oxyz$ , where axis  $x$  coincides with the centroidal axis, axes  $y$  and  $z$  are principal axes of the  
 175 cross section, and  $xz$  is the bending plane. Be  $\mathbf{x} = [x \ y \ z]^T$  the position vector and  $L$  the length of  
 176 the beam. For simplicity, a uniform cross section is considered. It is assumed that the material is  
 177 isotropic and linearly elastic.

178 Be  $\mathbf{u}(\mathbf{x}, t)$  the displacement vector,  $\mathbf{u}^T(\mathbf{x}, t) = [u_x \ u_y \ u_z]$ . According to the TM beam theory,  
 179 the small displacement components of a given point  $P(\mathbf{x})$  in the beam can be cast in the form

180

$$u_x(\mathbf{x}, t) = u(x, t) - z\varphi(x, t), \quad u_z(\mathbf{x}, t) = v(x, t), \quad u_y(\mathbf{x}, t) = 0 \quad (5a,b,c)$$

181

182 where, for a cross section at  $x$ ,  $u(x, t)$ ,  $v(x, t)$  and  $\varphi(x, t)$  denote the  $x$ -,  $z$ -displacement and the  
 183 rotation about the  $y$ -axis, the latter taken as positive if clockwise. The corresponding axial, bending  
 184 and shear strain components, as given by the small strain equations, are

185

$$\varepsilon(x, t) = \frac{\partial u(x, t)}{\partial x}, \quad \gamma(x, t) = \frac{\partial v(x, t)}{\partial x} - \varphi(x, t), \quad \chi(x, t) = -\frac{\partial \varphi(x, t)}{\partial x} \quad (6a,b,c)$$

186

187

188 **Local stress resultants**

189 Be  $\boldsymbol{\sigma}^{(l)}(\mathbf{x}, t) = [\sigma_x^{(l)} \quad \sigma_y^{(l)} \quad \sigma_z^{(l)} \quad \tau_{yz}^{(l)} \quad \tau_{xz}^{(l)} \quad \tau_{xy}^{(l)}]^T$  the vector of six components of the Cauchy stress  
190 tensor and be  $N^{(l)}(x, t)$ ,  $T^{(l)}(x, t)$  and  $M^{(l)}(x, t)$  the classical local stress resultants, i.e. normal  
191 stress, shear stress and bending moment given by

192

$$N^{(l)}(x, t) = \int_A \sigma_x^{(l)}(x, t) dA, \quad T^{(l)}(x, t) = \int_A \tau_{xz}^{(l)}(x, t) dA, \quad M^{(l)}(x, t) = \int_A \sigma_x^{(l)}(x, t) z dA \quad (7a,b,c)$$

193

194 The local stress resultants in Eqs.(7) are related to the corresponding axial, shear and bending strain  
195 by the constitutive laws of the TM beam:

196

$$N^{(l)}(x, t) = E^* A \varepsilon(x, t), \quad T^{(l)}(x, t) = K_s G^* A \gamma(x, t), \quad M^{(l)}(x, t) = E^* I \chi(x, t) \quad (8a,b,c)$$

197

198 where  $A$  and  $I$  are the area and the moment of inertia of the cross section,  $K_s$  is the shear  
199 correction factor,  $E^* = \beta_1 E$  and  $G^* = \beta_1 G$ , being  $E$  and  $G$  the Young and the shear modulus;  $\beta_1$  is  
200 a dimensionless coefficient,  $0 \leq \beta_1 \leq 1$ , that weighs the amount of local effects (Di Paola et al.  
201 2010b). In this respect, note that  $\beta_1$  is introduced here as in those non-local theories where the non-  
202 local material is conceived as a two-phase elastic material (Altan, 1989; Polizzotto, 2001).

203

204 **Long-range interactions**

205 Long-range interactions are modeled on a mechanical basis. The fundamental assumption is that  
206 two non-adjacent beam segments of volume  $\Delta V(x_i)$  and  $\Delta V(\xi_k)$  located, respectively, at  
207  $x = x_i$  and  $x = \xi_k$  on the beam axis, mutually exert long-range volume forces/moments as a  
208 result of their relative motion measured in terms of the “pure axial”, “pure bending” and “pure  
209 shear” deformation modes of a TM beam (Fuchs 1991, 1997). It is assumed that the long-range

210 volume forces/moments are self-equilibrated interactions, which counteract the relative motion of  
 211 the beam segments. The analytical form is built as linearly depending on the product of the volumes  
 212 of the interacting beam segments, through appropriate attenuation functions governing the space  
 213 decay of non-local effects. Purely elastic and fractional-order viscoelastic long-range volume  
 214 forces/moments are considered, the latter modeled by the Caputo's fractional derivative introduced  
 215 in Section 2. A mechanical description of the long-range interactions is shown in Figure 2.

216 In the pure axial deformation mode, two non-adjacent beam segments of volume  $\Delta V(x_i)$  and  
 217  $\Delta V(\xi_k)$  exchange long-range volume axial forces as a result of the relative axial displacement:

$$\eta(x_i, \xi_k, t) = u(\xi_k, t) - u(x_i, t) \quad (9)$$

218  
 219  
 220 The specific volume axial forces exchanged by unit volumes  $\Delta V(x_i) = 1$  and  $\Delta V(\xi_k) = 1$ , due  
 221 to the pure axial deformation (9), are given by

$$q_x(x_i, \xi_k, t) = r_x(x_i, \xi_k, t) + d_x(x_i, \xi_k, t) \quad (10)$$

$$r_x(x_i, \xi_k, t) = g_x(x_i, \xi_k) \eta(x_i, \xi_k, t) \Delta V(x_i) \Delta V(\xi_k) \quad (11)$$

$$d_x(x_i, \xi_k, t) = \tilde{g}_x(x_i, \xi_k) {}_C D_{0^+}^\alpha (\eta(x_i, \xi_k, t)) \Delta V(x_i) \Delta V(\xi_k) \quad (12)$$

222  
 223  
 224  
 225  
 226 Likewise, in the pure bending mode, two non-adjacent beam segments of volume  $\Delta V(x_i)$  and  
 227  $\Delta V(\xi_k)$  exchange long-range volume moments as a result of the relative rotation:

$$\theta(x_i, \xi_k, t) = \varphi(\xi_k, t) - \varphi(x_i, t). \quad (13)$$

228  
 229

230 In this case, the specific volume moments exchanged by  $\Delta V(x_i)=1$  and  $\Delta V(\xi_k)=1$  are given  
 231 as

$$q_{\varphi\varphi}(x_i, \xi_k, t) = r_{\varphi\varphi}(x_i, \xi_k, t) + d_{\varphi\varphi}(x_i, \xi_k, t) \quad (14)$$

232

$$r_{\varphi\varphi}(x_i, \xi_k, t) = g_{\varphi}(x_i, \xi_k) \theta(x_i, \xi_k, t) \Delta V(x_i) \Delta V(\xi_k) \quad (15)$$

233

$$d_{\varphi\varphi}(x_i, \xi_k, t) = \tilde{g}_{\varphi}(x_i, \xi_k) {}_C D_{0^+}^{\alpha} (\theta(x_i, \xi_k, t)) \Delta V(x_i) \Delta V(\xi_k) \quad (16)$$

234

235 Finally, in the pure shear mode, two non-adjacent beam segments of volume  $\Delta V(x_i)$  and  
 236  $\Delta V(\xi_k)$  exchange volume transverse forces and moments, as a result of their rotations with  
 237 respect to the line given by the relative transverse displacement, that is

238

$$\psi(x_i, \xi_k, t) = \left[ \frac{v(\xi_k, t) - v(x_i, t)}{\xi_k - x_i} - \varphi(\xi_k, t) \right] + \left[ \frac{v(\xi_k, t) - v(x_i, t)}{\xi_k - x_i} - \varphi(x_i, t) \right] \quad (17)$$

239

240 The specific volume transverse forces and moments exchanged by  $\Delta V(x_i)=1$  and  $\Delta V(\xi_k)=1$   
 241 are given by

242

$$q_z(x_i, \xi_k, t) = r_z(x_i, \xi_k, t) + d_z(x_i, \xi_k, t) \quad (18)$$

243

$$r_z(x_i, \xi_k, t) = \frac{2 \operatorname{sgn}(\xi_k - x_i)}{|x_i - \xi_k|} g_z(x_i, \xi_k) \psi(x_i, \xi_k, t) \Delta V(x_i) \Delta V(\xi_k) \quad (19)$$

244

$$d_z(x_i, \xi_k, t) = \frac{2 \operatorname{sgn}(\xi_k - x_i)}{|x_i - \xi_k|} g_z(x_i, \xi_k) {}_C D_{0^+}^{\alpha} (\psi(x_i, \xi_k, t)) \Delta V(x_i) \Delta V(\xi_k) \quad (20)$$

245

246 where obviously  $\frac{\operatorname{sgn}(\xi_k - x_i)}{|x_i - \xi_k|} = \frac{1}{\xi_k - x_i}$  and

$$q_{\varphi z}(x_i, \xi_k, t) = r_{\varphi z}(x_i, \xi_k, t) + d_{\varphi z}(x_i, \xi_k, t) \quad (21)$$

$$r_{\varphi z}(x_i, \xi_k, t) = g_z(x_i, \xi_k) \psi(x_i, \xi_k, t) \Delta V(x_i) \Delta V(\xi_k) \quad (22)$$

$$d_{\varphi z}(x_i, \xi_k, t) = \tilde{g}_z(x_i, \xi_k) {}_C D_{0^+}^\alpha(\psi(x_i, \xi_k, t)) \Delta V(x_i) \Delta V(\xi_k) \quad (23)$$

247

248

249

250 In Eqs.(12)-(16)-(20)-(23),  ${}_C D_{0^+}^\beta(\cdot)$  is the Caputo fractional derivative operator (1) as applied to  
 251 pure axial, pure bending and pure shear deformation modes.

252

### 253 **Remarks on the proposed model of long-range interactions**

254 In Eqs.(11)-(12) for the axial mode, Eqs.(15)-(16) for the bending mode, and Eqs.(19)-(20)-(22)-  
 255 (23) for the shear mode,  $g_s(x, \xi)$  and  $\tilde{g}_s(x, \xi)$ , for  $s = x, \varphi, z$ , are attenuation functions  
 256 governing the space decay of purely elastic and purely viscoelastic long-range interactions. They  
 257 shall be positive definite and must be taken as symmetric with respect to arguments  $x$  and  $\xi$ , to  
 258 ensure that the long-range resultants exchanged by the interacting beam segments are mutual,  
 259 according to Newton's third law. Further, notice that they are introduced as independent functions.  
 260 That is, by  $g_s(x, \xi) \neq \tilde{g}_s(x, \xi)$  for  $s = x, \varphi, z$ , a different spatial decay can be considered for  
 261 purely elastic and purely viscoelastic long-range interactions, while  $g_x(x, \xi) \neq g_\varphi(x, \xi) \neq g_z(x, \xi)$   
 262 and  $\tilde{g}_x(x, \xi) \neq \tilde{g}_\varphi(x, \xi) \neq \tilde{g}_z(x, \xi)$  mean that spatial decay may vary depending on pure axial,  
 263 pure bending and pure shear effects. This choice is made for the model to be as versatile as possible  
 264 for experimental data fitting. A possible choice could be adopting the same mathematical model for  
 265 the attenuation functions but with different parameters. Some examples of experimental data fitting  
 266 have been given by the authors, assuming the same exponential attenuation function for non-local  
 267 bending and shear effects in a non-local TM beam model (Alotta et al. 2014), or the same  
 268 exponential form but with different parameters for non-local bending and shear effects in a non-  
 269 local EB beam model (Di Paola et al. 2014). In both cases, purely elastic long-range interactions

270 proved capable of reproducing stiffening size effects in epoxy beams, measured experimentally by  
271 Lam et al. (2003). Attenuation functions alternative to exponential ones could be fractional power-  
272 law or Gaussian functions, for both elastic and viscoelastic non-local effects (Di Paola et al., 2009;  
273 Failla et al., 2011; Friswell et al. 2007). Notice that power-law decay of non-local effects is, indeed,  
274 the basic assumption of the fractional calculus approach to non-local elasticity (Tarasov and  
275 Zaslavsky 2007, 2008; Atanackovic and Stankovic 2009; Sapora et al. 2013; Carpinteri et al. 2014;  
276 Tarasov 2014; Tarasov and Aifantis 2015; Sumelka and Blaszczyk 2014).

277 The volume forces/moments (11)-(12), (15)-(16), (19)-(20) and (22)-(23) may model non-local  
278 effects of various nature, triggered by a differential motion. At a microstructural level, they could be  
279 thought as homogenized measures of inter-atomic interactions arising from bond-stretching and  
280 angle variation (Li and Chou 2003; Wan and Delale 2010). In this context, fractional viscoelastic  
281 long-range interactions could be suitable for modeling damping effects, as revealed by recent  
282 experiments (Payton et al. 2012, Murmu and Adhikari 2012, Calleja et al. 2012, Lee and Lin 2010,  
283 Chen et al. 2011). At a macroscopic scale, they could reflect viscoelastic forces transmitted from the  
284 fibers to the matrix in a composite beam with multi-oriented fiber reinforcements, with the  
285 viscoelastic modeling of the dissipation mechanism at the fiber-matrix interface (Gosz et al.,1991)  
286 accounting for imperfect bonding due to mechanical imperfections, unreacted polymer components,  
287 fiber treatments or, in some cases, for the presence of an “engineered” interphase between fibers  
288 and matrix, to optimize composite performances (Matzenmiller and Gerlach, 2004; Fisher and  
289 Brinson, 2001). Certainly, a quite interesting feature of the non-local model is the fact that separate  
290 pure axial, pure bending and pure shear long-range interactions can be accounted for. This makes  
291 the model suitable for those applications where it can be assumed that non-local effects result only  
292 in long-range moments but not in long-range transverse forces. This may be the case of particular  
293 microstructures or composite beams where longitudinal fibers passing through a material matrix are  
294 placed only at the upper and lower surface of the beam.

295 It is apparent that, when non-local stiffness terms (11)-(15)-(19)-(22) and non-local fractional-  
 296 order viscoelastic terms (12)-(16)-(20)-(23) are considered simultaneously in the model, the long-  
 297 range volume forces/moments can be interpreted as the result of a non-local fractional Kelvin-Voigt  
 298 connection between the interacting beam segments, see Figure 2. Obviously, when no viscoelastic  
 299 terms are considered, the model reverts to that presented by the authors in previous publications on  
 300 elastic non-local TM beams (Di Paola et al. 2014, Alotta et al. 2014, Failla et al. 2015).

301

### 302 NON-LOCAL BEAM MODEL EQUATIONS OF MOTION

303 Next, on dividing the beam in  $N$  segments of length  $\Delta x$ , the equations of motion of the beam  
 304 segment of volume  $\Delta V(x_i) = A\Delta x$  at  $x = x_i = i\Delta x$ , for  $i = 0, 1, \dots, N-1$  ( $x_0 = 0, x_N = x_L$ ), are written in  
 305 the form (see Figure 3):

306

$$N^{(l)}(x_i + \Delta x) - N^{(l)}(x_i) + Q_x(x_i, t) + F_x(x_i, t)\Delta x - m(x_i)\ddot{u}(x_i, t)\Delta x = 0 \quad (24a)$$

307

$$T^{(l)}(x_i + \Delta x) - T^{(l)}(x_i) + Q_z(x_i, t) + F_z(x_i, t)\Delta x - m(x_i)\ddot{v}(x_i, t)\Delta x = 0 \quad (24b)$$

308

$$M^{(l)}(x_i + \Delta x) - M^{(l)}(x_i) - T^{(l)}(x_i)\Delta x - Q_\varphi(x_i, t) + I_\rho(x_i)\ddot{\varphi}(x_i, t)\Delta x = 0 \quad (24c)$$

309

310 In Eqs.(24), dots mean differentiation with respect to time,  $F_x(x, t)$  and  $F_y(x, t)$  are introduced as  
 311 generalized measures per unit length of the external forces on the beam,  $m(x) = \rho(x)A$  and

312  $I_\rho(x) = \int_A \rho(x)z^2 dA$ , being  $\rho(x)$  the mass per unit volume. Eqs.(24) state that the equilibrium of

313 the beam segment of volume  $\Delta V(x_i)$ , at  $x = x_i$ , is attained due to the local stress resultants (7)

314 exerted by the adjacent beam segments, and the resultants  $Q_x$ ,  $Q_y$  and  $Q_\varphi$  of the volume

315 forces/moments exerted by all the non-adjacent beam segments of volume  $\Delta V(\xi_k)$  at  $x = \xi_k$ ,

316  $\xi_k \neq x_i$ , given as

317

$$\begin{aligned}
 Q_x(x_i, t) &= \sum_{k=0, k \neq i}^{N-1} q_x(x_i, \xi_k, t) \\
 Q_z(x_i, t) &= \sum_{k=0, k \neq i}^{N-1} q_z(x_i, \xi_k, t) \\
 Q_\varphi(x_i, t) &= \sum_{k=0, k \neq i}^{N-1} q_{\varphi\varphi}(x_i, \xi_k, t) + q_{\varphi z}(x_i, \xi_k, t)
 \end{aligned} \tag{25a-c}$$

318

319 For brevity,  $Q_x$ ,  $Q_y$  and  $Q_\varphi$  will be referred to as *long-range resultants*.

320 On replacing Eq.(10) for  $q_x$ , Eq.(14) for  $q_{\varphi\varphi}$ , Eq.(18) for  $q_z$  and Eq.(21) for  $q_{\varphi z}$ , dividing

321 Eqs.(24) by  $\Delta x$  and taking the limit  $\Delta x \rightarrow 0$  lead to the following equations:

322

$$E^* A \frac{\partial^2 u(x, t)}{\partial x^2} + F_x(x, t) + \tag{26a}$$

$$A^2 \int_0^L \left[ g_x(x, \xi) \eta(x, \xi, t) + \tilde{g}_x(x, \xi) {}_C D_{0^+}^\alpha (\eta(x, \xi, t)) \right] d\xi = m(x) \ddot{u}(x, t)$$

323

$$K_s G^* A \left[ \frac{\partial^2 v(x, t)}{\partial x^2} - \frac{\partial \varphi(x, t)}{\partial x} \right] + F_z(x, t) + \tag{26b}$$

$$A^2 \int_0^L \frac{2}{\xi - x} \left[ g_z(x, \xi) \psi(x, \xi, t) + \tilde{g}_z(x, \xi) {}_C D_{0^+}^\alpha (\psi(x, \xi, t)) \right] d\xi = m(x) \ddot{v}(x, t)$$

324

$$\begin{aligned}
& E^* I \frac{\partial^2 \varphi(x, t)}{\partial x^2} + K_s G^* A \left[ \frac{\partial v(x, t)}{\partial x} - \varphi(x, t) \right] + \\
& A^2 \int_0^L \left[ g_\varphi(x, \xi) \theta(x, \xi, t) + \tilde{g}_\varphi(x, \xi) {}_C D_{0^+}^\alpha (\theta(x, \xi, t)) \right] d\xi + \\
& A^2 \int_0^L \left[ g_z(x, \xi) \psi(x, \xi, t) + \tilde{g}_z(x, \xi) {}_C D_{0^+}^\alpha (\psi(x, \xi, t)) \right] d\xi = I_\rho(x) \ddot{\varphi}(x, t)
\end{aligned} \tag{26c}$$

325

326 where the constitutive local laws (8) have been introduced, and  $\Delta V(x) = A\Delta x$ ,  $\Delta V(\xi) = A\Delta \xi$  for  
327 the volumes of the interacting beam segments.

328 As for the boundary conditions (B.C.), it can readily be seen that the mechanical B.C. hold the  
329 classical form of local theory. This is true because, in the equilibrium equations at the beam ends,  
330 the long-range resultants (25) are infinitesimal of higher order with respect to the local stress  
331 resultants (e.g., see Di Paola et al. 2009). Also, time independent kinematic B.C. are considered.  
332 Therefore, the B.C. are given as

$$\begin{aligned}
E^* A \frac{\partial u(x, t)}{\partial x} \Big|_{x=x_i} &= \mp N_i(t), & \text{or} & \quad u(x_i, t) = u_i \\
K_s G^* A \left[ \frac{\partial v(x, t)}{\partial x} - \varphi(x, t) \right] \Big|_{x=x_i} &= \mp T_i(t), & \text{or} & \quad v(x_i, t) = v_i \\
E^* I \frac{\partial \varphi(x, t)}{\partial x} \Big|_{x=x_i} &= \mp M_i(t), & \text{or} & \quad \varphi(x_i, t) = \varphi_i
\end{aligned} \tag{27a-c}$$

333

334 where  $N_i$ ,  $M_i$  and  $T_i$ ,  $u_i$ ,  $v_i$  and  $\varphi_i$ , denote the external forces/moments,  
335 displacements/rotations at the beam ends, i.e. at  $x_0 = 0$  and  $x_L = L$ .

336 The equilibrium equations (26) clearly show that the non-local beam model is a displacement-  
337 based model, with long-range volume forces/moments that arise from relative  
338 displacements/rotations between non-adjacent beam segments, as given by the pure deformation  
339 modes (9)-(13)-(17). On the contrary, if the long-range volume transverse forces/moments were

340 taken as depending on the relative transverse displacement and not on the pure shear deformation  
341 (17), long-range volume transverse forces/moments would erroneously arise from a relative  
342 transverse displacement induced, for instance, by a rigid rotation of the beam. That is, the non-local  
343 beam model is invariant with respect to rigid body motion and axial, bending and shear non-local  
344 behaviors are mechanically consistent.

345 The integral terms on the l.h.s. of Eqs.(26) are the long-range resultants per unit length.  
346 Interestingly, the viscoelastic long-range axial force in Eq.(26a) and moment in Eq.(26b),  
347 specifically the part due to the pure bending deformation mode (13), correspond to those introduced  
348 by Russell (1992) in his non-local damping model for a bar and a EB composite beam with  
349 longitudinal embedded fibers. Unlike the model proposed by Russell (1992), however, the proposed  
350 model includes long-range transverse forces/moments due to the asymmetric “pure shear”  
351 deformation mode between non-adjacent beam segments, and mechanical B.C. identical to those of  
352 classical local theory.

353 Finally, recognize that the non-local damping model is not proportional, as the fractional-order  
354 viscoelastic terms do not have the analytical form of the elastic ones, to which contribute both local  
355 and non-local terms.

356

## 357 **FINITE ELEMENT FORMULATION**

358 Following a standard approach of the FE method, consider a mesh with  $n$  disjointed elements of the  
359 same length, along the beam axis. Points shared by contiguous elements are *mesh nodes*. Abscissas  
360 of the nodes of the  $i^{\text{th}}$  element are denoted as  $\hat{x}_i$  and  $\hat{x}_{i+1}$ , with  $\hat{x}_1 = 0$  and  $\hat{x}_{n+1} = L$  (symbol “^” is  
361 introduced to avoid confusion with abscissas  $x_i$ ’s used in Sections 3-4), and  $l$  denotes the length of  
362 the  $i^{\text{th}}$  element. The displacement field within the  $i^{\text{th}}$  element is given the following form

363

$$\mathbf{u}_i(x, t) = \mathbf{N}_i(x) \mathbf{d}_i(t) \quad i = 1, 2, \dots, n \quad (28)$$

364

365 In Eq.(28),  $\mathbf{u}_i(x, t) = [u(x, t) \quad v(x, t) \quad \varphi(x, t)]^T$  is the vector of displacements/rotation  
 366 within the  $i^{\text{th}}$  element,  $\mathbf{d}_i(t)$  is the vector of the unknown nodal displacements of the  $i^{\text{th}}$  element,  
 367 i.e.

$$\mathbf{d}_i(t) = [u_{(i)1}(t) \quad v_{(i)1}(t) \quad \varphi_{(i)1}(t) \quad u_{(i)2}(t) \quad v_{(i)2}(t) \quad \varphi_{(i)2}(t)]^T \quad (29)$$

369  
 370 where subscript “(i)” indicates the  $i^{\text{th}}$  element, while subscripts 1-2 denote first and second node of  
 371 the element. In Eq.(28),  $\mathbf{N}_i(x)$  is the matrix collecting the shape functions taken, in this paper, as  
 372 the standard 1<sup>st</sup> order and 3<sup>rd</sup> order polynomial shape functions of the two-node TM beam element,  
 373 for the axial and flexural response respectively. That is,  $\mathbf{N}_i(x)$  is given as

$$\mathbf{N}_i^T(x) = \begin{bmatrix} \frac{\hat{x}_{i+1} - x}{l} & 0 & 0 \\ 0 & \frac{(l - y_i)(l^2(1 + 12\Omega) + (l - 2y_i)y_i)}{l^3(1 + 12\Omega)} & \frac{6y_i(-l + y_i)}{l^3(1 + 12\Omega)} \\ 0 & \frac{(l - y_i)(l + 6l\Omega - y_i)y_i}{l^2(1 + 12\Omega)} & \frac{(l + 12l\Omega - 3y_i)(l - y_i)}{l^2(1 + 12\Omega)} \\ \frac{x - \hat{x}_i}{l} & 0 & 0 \\ 0 & \frac{y_i(12l^2\Omega + 3ly_i - 2y_i^2)}{l^3(1 + 12\Omega)} & \frac{6(l - y_i)y_i}{l^3(1 + 12\Omega)} \\ 0 & -\frac{(l - y_i)y_i(6l\Omega + y_i)}{l^2(1 + 12\Omega)} & \frac{y_i(2l(-1 + 6\Omega) + 3y_i)}{l^2(1 + 12\Omega)} \end{bmatrix} \quad (30)$$

374  
 375 where  $y_i = x - \hat{x}_i$  and  $\Omega = E^* I / G^* A l^2$ .

376 Being  $\mathbf{d} = [u_1 \quad v_1 \quad \varphi_1 \quad u_2 \quad v_2 \quad \varphi_2 \quad \dots \quad u_n \quad v_n \quad \varphi_n]^T$  the vector collecting all nodal displacements of  
 377 the mesh, the nodal displacements of the  $i^{\text{th}}$  element are written as

$$\mathbf{d}_i(t) = \mathbf{C}_i \mathbf{d}(t) \quad (31)$$

379

380 being  $\mathbf{C}_i$  the connectivity matrix. Next, following a standard Galerkin approach, the following  
 381 equations can be derived

$$\mathbf{M}\ddot{\mathbf{d}}(t) + \mathbf{C}^{(nl)} \left( {}_c D_{0^+}^\alpha \mathbf{d}(t) \right) + \mathbf{K}\mathbf{d}(t) = \mathbf{F}(t) \quad (32)$$

382  
 383  
 384 In Eq.(32),  $\mathbf{K}$  is the  $3(n+1) \times 3(n+1)$  global stiffness matrix, given as

$$\mathbf{K} = \mathbf{K}^{(l)} + \mathbf{K}^{(nl)} = \sum_{i=1}^n \mathbf{K}_i^{(l)} + \sum_{i=1}^n \mathbf{K}_i^{(nl)} \quad (33)$$

385  
 386  
 387 where  $\mathbf{K}_i^{(l)}$  and  $\mathbf{K}_i^{(nl)}$  are local and non-local stiffness matrices, respectively. The first is given as

$$\mathbf{K}_i^{(l)} = A \int_{\hat{x}_i}^{\hat{x}_{i+1}} (\mathbf{B}_i(x) \mathbf{C}_i)^T \mathbf{D}^* \mathbf{B}_i(x) \mathbf{C}_i dx \quad (34)$$

388  
 389  
 390 where  $\mathbf{D}^* = \text{Diag} [E^* A \quad E^* I \quad G^* K_S A]$ ,  $\mathbf{B}_i(x)$  is the  $3 \times 6$  matrix

$$\mathbf{B}_i^T(x) = \begin{bmatrix} -\frac{1}{l} & 0 & 0 \\ 0 & \frac{6(-2l^2\Omega - ly_i + y_i^2)}{l^3(1+12\Omega)} & -\frac{6(l-2y_i)}{l^3(1+12\Omega)} \\ 0 & \frac{l^2(1+6\Omega) - 4(l+3l\Omega)y_i + 3y_i^2}{l^2(1+12\Omega)} & \frac{-4(l+3l\Omega) + 6y_i}{l^2(1+12\Omega)} \\ \frac{1}{l} & 0 & 0 \\ 0 & \frac{6(2l^2\Omega + (l-y_i)y_i)}{l^3(1+12\Omega)} & \frac{6(l-2y_i)}{l^3(1+12\Omega)} \\ 0 & \frac{-6l^2\Omega + 2l(-1+6\Omega)y_i + 3y_i^2}{l^2(1+12\Omega)} & \frac{2l(-1+6\Omega) + 3y_i}{l^2(1+12\Omega)} \end{bmatrix} \quad (35)$$

391  
 392

393 being  $y_i = x - \hat{x}_i$ . The second is given as

394

$$\mathbf{K}_i^{(nl)} = \mathbf{K}_i^{(nl,\eta)} + \mathbf{K}_i^{(nl,\theta)} + \mathbf{K}_i^{(nl,\psi)} = \sum_{j=1}^n \mathbf{K}_{ij}^{(nl,\eta)} + \sum_{j=1}^n \mathbf{K}_{ij}^{(nl,\theta)} + \sum_{j=1}^n \mathbf{K}_{ij}^{(nl,\psi)} \quad (36)$$

395

396 In Eq.(36), matrices  $\mathbf{K}_{ij}^{(nl,\eta)}$ ,  $\mathbf{K}_{ij}^{(nl,\theta)}$ ,  $\mathbf{K}_{ij}^{(nl,\psi)}$  include the non-local stiffness contributions due to the

397 long-range interactions between the between the differential volumes  $dV(x) = A dx$  inside the  $i^{\text{th}}$

398 element  $(\hat{x}_i \leq x \leq \hat{x}_{i+1})$ , and the differential volumes  $dV(\xi) = A d\xi$  inside the  $j^{\text{th}}$  element

399  $(\hat{x}_j \leq \xi \leq \hat{x}_{j+1})$ , namely

$$\mathbf{K}_{ij}^{(nl,\eta)} = \frac{A^2}{2} \int_{\hat{x}_i}^{\hat{x}_{i+1}} \int_{\hat{x}_j}^{\hat{x}_{j+1}} (\mathbf{N}_j^{(u)}(\xi) \mathbf{C}_j - \mathbf{N}_i^{(u)}(x) \mathbf{C}_i)^T \mathbf{g}_x(x, \xi) (\mathbf{N}_j^{(u)}(\xi) \mathbf{C}_j - \mathbf{N}_i^{(u)}(x) \mathbf{C}_i) dx d\xi \quad (37a)$$

400

$$\mathbf{K}_{ij}^{(nl,\theta)} = \frac{A^2}{2} \int_{\hat{x}_i}^{\hat{x}_{i+1}} \int_{\hat{x}_j}^{\hat{x}_{j+1}} (\mathbf{N}_j^{(\phi)}(\xi) \mathbf{C}_j - \mathbf{N}_i^{(\phi)}(x) \mathbf{C}_i)^T \mathbf{g}_\phi(x, \xi) (\mathbf{N}_j^{(\phi)}(\xi) \mathbf{C}_j - \mathbf{N}_i^{(\phi)}(x) \mathbf{C}_i) dx d\xi \quad (37b)$$

401

$$\mathbf{K}_{ij}^{(nl,\psi)} = \frac{A^2}{2} \int_{\hat{x}_i}^{\hat{x}_{i+1}} \int_{\hat{x}_j}^{\hat{x}_{j+1}} \left( 2 \frac{\mathbf{N}_j^{(v)}(\xi) \mathbf{C}_j - \mathbf{N}_i^{(v)}(x) \mathbf{C}_i}{\xi - x} - \mathbf{N}_j^{(\phi)}(\xi) \mathbf{C}_j - \mathbf{N}_i^{(\phi)}(x) \mathbf{C}_i \right)^T \mathbf{g}_z(x, \xi) \left( 2 \frac{\mathbf{N}_j^{(v)}(\xi) \mathbf{C}_j - \mathbf{N}_i^{(v)}(x) \mathbf{C}_i}{\xi - x} - \mathbf{N}_j^{(\phi)}(\xi) \mathbf{C}_j - \mathbf{N}_i^{(\phi)}(x) \mathbf{C}_i \right) dx d\xi \quad (37c)$$

402

403 In Eqs.(37),  $\mathbf{N}_i^{(u)}$ ,  $\mathbf{N}_i^{(v)}$  and  $\mathbf{N}_i^{(\phi)}$  are row vectors of the shape functions matrix  $\mathbf{N}_i$ , i.e.

404

$$\mathbf{N}_i^{(u)T}(x) = \frac{1}{l} \begin{bmatrix} \hat{x}_{i+1} - x \\ 0 \\ 0 \\ x - \hat{x}_i \\ 0 \\ 0 \end{bmatrix} \quad (38a)$$

405

$$\mathbf{N}_i^{(v)T}(x) = \frac{1}{l^3(1+12\Omega)} \begin{bmatrix} 0 \\ (l-y_i)(l^2(1+12\Omega)+(l-2y_i)y_i) \\ l(l-y_i)(l+6l\Omega-y_i)y_i \\ 0 \\ y_i(12l^2\Omega+3ly_i-2y_i^2) \\ -l(l-y_i)y_i(6l\Omega+y_i) \end{bmatrix} \quad (38b)$$

406

$$\mathbf{N}_i^{(\phi)T}(x) = \frac{1}{l^3(1+12\Omega)} \begin{bmatrix} 0 \\ 6y_i(-l+y_i) \\ l(l+12l\Omega-3y_i)(l-y_i) \\ 0 \\ 6(l-y_i)y_i \\ ly_i(2l(-1+6\Omega)+3y_i) \end{bmatrix} \quad (38c)$$

407

408 being  $y_i = x - \hat{x}_i$ . Further, in Eq.(32) matrix  $\mathbf{C}^{(nl)}$  is the  $3(n+1) \times 3(n+1)$  global viscoelastic matrix. It

409 is easy to recognize that  $\mathbf{C}^{(nl)}$  has the same mathematical form as the non-local stiffness matrix

410  $\mathbf{K}^{(nl)}$  where, however,  $g_s(x, \xi)$  are replaced by  $\tilde{g}_s(x, \xi)$ , for  $s = x, \phi, z$ . Further, in Eq.(32) matrix

411  $\mathbf{M}$  is the  $3(n+1) \times 3(n+1)$  global consistent mass matrix (Reddy, 2006), while vector  $\mathbf{F}(t)$  is the load

412 vector given as

413

$$\mathbf{F}(t) = \sum_{i=1}^n \mathbf{F}_i(t) \quad (39)$$

414

415 with

$$\mathbf{F}_i(t) = \int_{V_i} (\mathbf{N}_i(x) \mathbf{C}_i)^T \bar{\mathbf{F}}(x, t) dV_i(x) + (\mathbf{N}_i(0) \mathbf{C}_i)^T \bar{\mathbf{F}}_0(t) + (\mathbf{N}_i(L) \mathbf{C}_i)^T \bar{\mathbf{F}}_L(t) \quad (40)$$

416

417 being  $\bar{\mathbf{F}}(x, t) = [F_x(x, t) \quad F_z(x, t) \quad 0]^T$ ,  $\bar{\mathbf{F}}_i(t) = [N_i(t) \quad T_i(t) \quad M_i(t)]^T$ ,  $i = 0, L$ .

418 Finally, two important remarks are in order. Unlike the local stiffness matrix  $\mathbf{K}^{(l)}$ , the non-local  
419 stiffness matrix  $\mathbf{K}^{(nl)}$  and the viscoelastic matrix  $\mathbf{C}^{(nl)}$  are fully-populated. Also, closed-form  
420 solutions for the elements of  $\mathbf{K}^{(nl)}$  and  $\mathbf{C}^{(nl)}$  can be obtained for attenuation functions  $g_s(x, \xi)$   
421 and  $\tilde{g}_s(x, \xi)$  of common use in non-local theories, such as exponential or power-law functions.  
422 Details can be found in a previous study by the authors (Alotta et al. 2014) and are not reported  
423 here, for brevity.

424

## 425 TIME-DOMAIN SOLUTION

426 Given an arbitrary input  $\mathbf{F}(t)$ , Eq.(32) can be solved in the time domain following a general  
427 approach (Bagley and Torvik 1985; Bagley and Calico 1991; Di Paola and Pinnola 2014), which is  
428 based on the complex eigensolution of a multi-degree-of-freedom companion system, obtained from  
429 Eq.(32) by including a suitable number of additional state variables. However, load cases of  
430 particular interest, such as creep tests, can be tackled by closed-form solutions, as explained in the  
431 following.

432 In a typical creep test, it can be assumed that the load vector  $\mathbf{F}(t)$  in Eq.(32) attains a constant  
433 value  $\mathbf{F}(t) = \mathbf{F}$  at a given time after  $t = t_0$ , with a slow initial loading rate. Under this assumption,  
434 inertial terms in Eq.(32) can be neglected and, being  $\Phi$  the eigenvectors matrix of  $\mathbf{A} = \mathbf{K}^{-1}\mathbf{C}_{NL}$ ,  
435 Eq.(32) can be recast as follows

436

$$\Lambda({}_c D_{0^+}^\alpha \mathbf{z})(t) + \Omega \mathbf{z}(t) = \Phi^T \mathbf{F}(t) \quad (41)$$

437

438 where  $\mathbf{z}(t) = \Phi^{-1} \mathbf{d}(t)$ , while  $\Lambda = \Phi^T \mathbf{C}^{(nl)} \Phi$  and  $\Omega = \Phi^T \mathbf{C}^{(nl)} \Phi$  are diagonal matrices. System  
439 (41) is uncoupled. For instance, if the load vector  $\mathbf{F}(t)$  is given the analytical form

440  $\mathbf{F}(t) = \mathbf{F} \cdot t/t_0 + \mathbf{F} \cdot U(t-t_0)(1-t/t_0)$ , with  $U(t-t_0)$  denoting the unit-step function, exact closed-  
 441 form solutions for components can be obtained for load cases of particular interest in  
 442 viscoelasticity. For a typical creep test under a constant load distributed over the beam,  
 443  $\mathbf{F}(t) = \mathbf{F} \cdot U(t)$ , where  $U(t)$  is the unit-step function, it yields

$$\lambda_j \left( {}_c D_{0^+}^\alpha z_j(t) \right) + \omega_j z_j(t) = f_j \cdot t/t_0 + f_j \cdot U(t-t_0)(1-t/t_0) \quad (42)$$

444  
 445  
 446 where  $f_j = \Phi_j^T \mathbf{F}$ ,  $\lambda_j$  and  $\omega_j$  denote the  $j$ th elements of matrices  $\Lambda$  and  $\Omega$ , while  $z_j(t)$  is given  
 447 by a Mittag-Leffler function, as follows

$$z_j(t) = \frac{t}{t_0 \omega_j} \left[ 1 - E_{\alpha,2} \left( -\frac{\omega_j}{\lambda_j} t^\alpha \right) \right] - \frac{(t-t_0)U(t-t_0)}{t_0 \omega_j} \left[ 1 - E_{\alpha,2} \left( -\frac{\omega_j}{\lambda_j} (t-t_0)^\alpha \right) \right] \quad (43)$$

448  
 449  
 450 for

$$E_{\alpha,\gamma}(w) = \sum_{k=0}^{\infty} \frac{w^k}{\Gamma(\alpha k + \gamma)} \quad (44)$$

## 451 452 NUMERICAL APPLICATIONS

453 The behavior of the non-local TM beam model is illustrated focusing on the flexural response.  
 454 Theoretical creep response of a simply-supported epoxy micro-beam with rectangular cross section  
 455 will be presented, for the following parameters: Young's modulus  $E=1.40$  GPa, Poisson's  
 456 coefficient  $\nu=0.35$ ;  $L=300$   $\mu\text{m}$ ,  $b=30$   $\mu\text{m}$  and  $h=15$   $\mu\text{m}$  are length, width and thickness of the cross  
 457 section. In the local constitutive equations (8),  $\beta_1=1$  is selected. As for the long-range interactions,  
 458 it is assumed that pure bending and shear behaviors are governed by the same attenuation functions,  
 459 i.e.  $g_s(x, \xi) = g(x, \xi)$ ,  $\tilde{g}_s(x, \xi) = \tilde{g}(x, \xi)$  for  $s = x, z, \varphi$ , with the following exponential forms:

460

$$g(x, \xi) = \frac{C}{h^2} \exp(-|x - \xi|/\lambda) \quad (45a)$$

$$\tilde{g}(x, \xi) = \frac{C_\alpha}{h^2} \exp(-|x - \xi|/\lambda_\alpha) \quad (45b)$$

461

462

463 where  $\lambda$  and  $\lambda_\beta$  are internal lengths. The non-local parameters  $(C, \lambda)$  and  $(C_\alpha, \lambda_\alpha)$  in Eqs.(45) are set  
 464 in order to enhance non-local effects and assess how they affect the response. The larger is the  
 465 internal length, the wider is the so-called influence distance, i.e. the maximum distance beyond  
 466 which the attenuation functions and therefore the non-local effects become negligible. Notice that,  
 467 as a result of the choice  $\beta_1 = 1$ , the non-local solution will tend to the solution obtained by the  
 468 classical local TM theory, as  $\lambda \rightarrow 0$  in Eq.(45a) and  $\lambda_\alpha \rightarrow 0$  in Eq.(45b).

469 Using Eqs.(45) for the attenuation functions  $g_s(x, \xi)$  and  $\tilde{g}_s(x, \xi)$ , for  $s = x, z, \varphi$ , terms in the  
 470 non-local stiffness matrix  $\mathbf{K}^{(nl)}$  and viscoelastic matrix  $\mathbf{C}^{(nl)}$  can be built in a closed form (Alotta et  
 471 al. 2014). Upon discretizing the equations of motion by the FE method, time-domain closed-form  
 472 solutions are built based on Eqs.(41)-(42) in Section 6.

473 Consider the uniformly-distributed load

474

$$p(t) = p_0 \cdot t/t_0 + p_0 \cdot U(t-t_0)(1-t/t_0); \quad p_0 = 1 \text{ Nm}^{-1}, \quad t_0 = 10 \text{ s} \quad (46)$$

475

476 Figure 4 through Figure 7 show the beam deflection as time elapses, normalized to the midspan  
 477 deflection of the classical TM beam theory,  $v^{(l)}(L/2)$ , when 40 FEs are used. The first relevant  
 478 observation is that the proposed model is capable of providing a large variety of viscoelastic  
 479 behaviors as the fractional order  $\alpha$  varies. This is a typical feature of fractional viscoelastic models,  
 480 representing a significant advantage compared to classical viscoelastic models that combine  
 481 multiple Maxwell or Kelvin-Voigt elements, as they generally involve a large number of  
 482 parameters. It is also seen that, regardless of  $\alpha$ , the deflection tends to the purely elastic non-local

483 one, more rapidly as the fractional order  $\alpha$  increases. It is worth noticing that the non-local  
484 deflection is stiffer than the corresponding classical local one, as a result of the stiffening effects  
485 due to the elastic long-range interactions, which counteract the relative motion between non-  
486 adjacent beam segments and, as such, provide additional stiffness with respect to the stiffness of the  
487 classical local TM terms (indeed,  $\beta_1 = 1$  has been set in the local constitutive equations (8)).  
488 Solutions with a larger number of FEs do not differ from the ones shown in Figures 4-7, and are not  
489 reported for clarity.

490 Figure 8 shows the midspan deflection at given time instants, normalized to the midspan  
491 deflection of the classical TM beam theory,  $v^{(l)}(L/2)$ . Consistently with Figures 4-7, the midspan  
492 deflection tends to the purely elastic non-local counterpart as time elapses, and more rapidly as the  
493 fractional order  $\alpha$  increases.

494 For a further insight into the proposed model, Figure 9 and Figure 10 show the midspan  
495 deflection for  $\alpha = 0.5$ , as parameters  $(C_\alpha, \lambda_\alpha)$  in Eq.(45b) vary (again, normalized to the midspan  
496 deflection of the classical TM beam theory,  $v^{(l)}(L/2)$ ). In particular,  $\lambda_\alpha = \text{cost}$  and  $C_\alpha$  varies in  
497 Figure 9, while  $C_\alpha = \text{cost}$  while  $\lambda_\alpha$  varies in Figure 10. It is clear that viscoelastic effects do  
498 increase with increasing  $C_\alpha$  in Figure 9 and increasing  $\lambda_\alpha$  in Figure 10. These results are consistent  
499 with the fact that, while  $C_\alpha$  governs the magnitude of the viscoelastic long-range interactions,  $\lambda_\alpha$   
500 governs the distance beyond which such interactions are negligible and, consequently, the number  
501 of beam segments interacting with a given one.

502

## 503 CONCLUDING REMARKS

504 A non-local TM beam with fractional-order viscoelastic long-range interactions has been  
505 presented, within the theoretical framework of a recent mechanically-based approach to non-local  
506 theory (Di Paola et al. 2009, 2010a, 2010b; Failla et al. 2010, 2013). The key assumption is that

507 classical stress resultants and long-range resultants contribute to the equilibrium of every beam  
508 segment. Classical stress resultants are exerted by adjacent segments, and long-range resultants are  
509 exchanged with all non-adjacent beam segments, as a result of relative motion measured by the pure  
510 deformation modes of TM beam kinematics, i.e. pure axial, bending and shear deformation modes.  
511 The long-range resultants are constructed as volume forces/moments, linearly-depending on the  
512 product of the volumes of the interacting beam segments through space-dependent attenuation  
513 functions, as is typical in non-local continua. Elastic and fractional-order viscoelastic long-range  
514 interactions are considered in the model. While the first depend on the pure deformation modes, the  
515 second depend on Caputo's fractional derivatives (Podlubny 1999) of the pure deformation modes.  
516 The resulting equilibrium equations are fractional differential equations. Since the long-range  
517 resultants are built as volume forces/moments, it is found that the related B.C. coincide with those  
518 of classical local theory (Di Paola et al. 2009). The model can be considered as a generalization of  
519 previous non-local models proposed by the authors (Di Paola et al. 2013, Di Paola et al. 2014,  
520 Alotta et al. 2014), which included purely elastic or Kelvin-Voigt viscoelastic long-range  
521 interactions.

522 The FE method has been applied to discretize the equilibrium equations. FE equations involve  
523 the classical local stiffness matrix  $\mathbf{K}^{(l)}$ , and non-local stiffness and viscoelastic matrices  $\mathbf{K}^{(nl)}$  and  
524  $\mathbf{C}^{(nl)}$  associated with the long-range resultants. For typical creep tests, Mittag-Leffler power series  
525 closed-form solutions (Podlubny 1999) have been built based on a suitable representation of the  
526 displacement response, based on the eigenvectors of the matrix given as the product between the  
527 inverse of the global stiffness matrix  $\mathbf{K} = \mathbf{K}^{(l)} + \mathbf{K}^{(nl)}$  and the non-local viscoelastic matrix  $\mathbf{C}^{(nl)}$ .  
528 Numerical applications have investigated the creep response of a simply-supported beam under a  
529 uniform load, assuming a typical exponential form for the spatial decay of elastic and viscoelastic  
530 long-range interactions (Friswell et al. 2007). Parameters and attenuation functions have been set on  
531 a theoretical basis, to enhance non-local effects. Results have shown that the model is quite

532 versatile, and potentially capable of providing a large variety of viscoelastic responses with a  
533 limited number of parameters, as is typical in fractional modeling of viscoelasticity as compared  
534 with traditional viscoelastic models combining Maxwell and Kelvin-Voigt models.

535 The proposed model shares with alternative non-local beam models the idea of a continuum  
536 enriched with non-local terms. The mathematical form assumed for the non-local terms, i.e. long-  
537 range volume forces/moments acting on every beam volume as a result of its interaction with non-  
538 adjacent beam volumes, appears consistent with the typical approach of engineering beam theories,  
539 where the equilibrium of a beam segment is set in an average (weak) sense based on the stress  
540 resultants on the cross section (normal and shear forces, bending moment). Applications may be  
541 envisaged to capture non-local damping effects due to micro-structural effects, as well as those  
542 arising, for instance, at the fiber-matrix interface in fiber-reinforced composite beams.

543 Further developments will focus on appropriate mathematical treatment of the fractional  
544 viscoelastic response when uncertainty is considered (Muscolino et al. 2013).

545

#### 546 **ACKNOWLEDGEMENTS**

547 PRIN 2010-2011: Stability, Control and Reliability of Flexible Structures”, National Coordinator  
548 Prof. A. Luongo, is gratefully acknowledged.

549

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